

SERVICE BULLETIN

CHANGE OF COOLANT SPECIFICATION ON ROTAX® ENGINE TYPE 912 AND 914 (SERIES) SB-912-043 R1 SB-914-029 R1

MANDATORY

Repeating symbols:

Please, pay attention to the following symbols throughout this document emphasizing particular information.

- ▲ WARNING: Identifies an instruction, which if not followed, may cause serious injury or even death.
- CAUTION: Denotes an instruction which if not followed, may severely damage the engine or could lead to
 - suspension of warranty.
- ◆ NOTE: Information useful for better handling.

1) Planning information

1.1) Engines affected

All versions of the engine type:

- 912 A all
- 912 F all
- 912 S all
- 914 F all
- all aircraft manufacturer which aircraft equipped with ROTAX_® engine type as listed.

1.2) Concurrent ASB/SB/SI and SL

More to this Service Bulletin the following additional Service Instruction must be observed and complied with:

- SI-912-016 / SI-914-019 "Selection of suitable operating fluids" current issue

1.3) Reason

Using conventional coolant with a mixing ratio of 50% coolant and 50% water may cause in some applications boiling of the coolant before reaching the max. allowable cylinder head temperature.

On all affected engines the following changes have to be considered and accomplished (if applicable).

- Change of coolant specification
- Change of the cylinder head temperature limits
- Introduction of a new coolant temperature limit and the requirement to measure and monitor this parameter
- Change of radiator cap

1.4) Subject

Change of coolant specification on ROTAX_® engine type 912 (Series) and 914 (Series)

1.5) Compliance

- at latest December 31, 2007, incorporate the mandatory use of waterless coolant into the relevant documentation of the aircraft.

Alternatively after this time limit the use of conventional coolant is possible. In such case the new operating limit (coolant temperatur) has to be applied. The work/compliance has to be performed according to section 3.

▲ WARNING: Non-compliance with these instructions could result in engine damages, personal injuries or death.

1.6) Approval

The technical content is approved under the authority of DOA Nr. EASA.21J.048.

1.7) Manpower

Engine installed in the aircraft - - - manpower time will depend on installation and thus, no estimate is available from the engine manufacturer.

1.8) Mass data

Change of weight - - - none

Moment of inertia - - - unaffected

1.9) Electrical load data

No change

1.10) Software accomplishment summary

No change

1.11) References

In addition to this technical information refer to current issue of

- Operator's Manual (OM)
- Illustrated Parts Catalog (IPC)
- Maintenance Manual (MM)
- all relevant Service Instructions (SI)

1.12) Other publications affected

The following documentations become effective with this Service Bulletin:

Description	part no.	Issue	Date	Rev.	Chapter	Page
Operator's Manual 912 Series	899370	0	1998 07 01	4*		
Operator's Manual 914 Series	899641	0	1998 12 01	4*		
Installation Manual 912 A	897860	0	1997 01 16	2*		
Installation Manual 912 F	897796	0	1996 01 23	3*		
Installation Manual 912 S	899376	0	1998 09 01	2*		
Installation Manual 912 UL	897711	2	1997 03 26	2*		
Installation Manual 914 F	897816	1	2006 02 01	0*		
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^{*} or higher revision

The replacement pages have to be included without delay into the respective documentation of the aircraft manufacturer.

1.13) Interchangeability of parts

Not affected

2) Material Information

2.1) Material - cost and availability

Price and availability will be supplied on request by ROTAX, Authorized Distributors or their Service Centers.

2.2) Company support information

None

2.3) Material requirement per engine

Fig.no.	New part no.	Qty/engine	Description	Old part no.	Application
	922075	1	radiator cap	922070*	ROTAX _® 912/914 (Series)
* if not in	nstalled already	See SI-25-19	97 Running Modific	cations" current issue	ů .

2.4) Rework of parts

None

2.5) Special tooling/lubricant-/adhesives-/sealing compound - Price and availability

None

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3) Accomplishment/Instructions

Accomplishment

3.1) Replacement of the radiator cap

Replace on all affected engines the radiator cap part no. 922075 (0,9 bar) (13 psi) with a new radiator cap part no. 922070 (1,2 bar) (18 psi).

♦ NOTE: The boiling point depends additionally on other factors such as the system pressure. At higher system pressure slightly higher boiling point can be obtained.

All work has to performed in accordance with the relevant Maintenance Manual.

3.2) Check cooling system - Efficiency of the cooling system

These measures must be performed by the aircraft manufacturer.

♦ NOTE: All work has to performed in accordance with the relevant Installation Manual (section Cooling system).

3.2.1) Determination of the achievable maximum coolant temperature and cylinder head temperature Depending on the maximum operating temperature achieved following measures have to be taken:

maximum values for		coolant used for tests		
Coolant temperature	Cylinder head temperature	Conventional coolant	Waterless coolant	
less than 120 °C (248 °F) 150 °C ¹⁾ (266 °F) 150 °C ¹⁾ (300 °F)		Additional instruments for displaying coolant temperature is necessary 2)	Modifications to the instruments	
more than 120 °C (248 °F)	less than 135 °C ¹⁾ (266 °F) 150 °C ¹⁾ (300 °F)		or limit not necessary	
less than 120 °C (248 °F) more than 135 °C ¹⁾ (266 °F) 150 °C ¹⁾ (300 °F)		Cooling capacity too low. Check of the installation necessary	Cooling capacity too low. Check	
more than 120 °C (248 °F)	more than 135 °C ¹⁾ (266 °F) 150 °C ¹⁾ (300 °F)		of the installation necessary	

¹⁾ depending on the engine type

◆ NOTE: Depending on the gained temperatures (coolant outlet temperature compared to cylinder head temperature) the aircraft manufacturer has to adapt the obtained limits in the relevant documentation (operators manual and/or flight manual).

Approval of translation to best knowledge and judgment - in any case the original text in the German language and the metric units (SI-system) are authoritative.

²⁾ monitoring of the coolant temperature by means of the cylinder head temperature is possible. See current relevant Installation Manual.