

SERVICE LETTER

Information about storage and storage conditions of ROTAX® genuine spare parts for ROTAX® Aircraft Engines

ATA System: 10-00-00 Storage and Installation

1) Planning information

To obtain satisfactory results, procedures specified in this publication must be accomplished with accepted methods in accordance with prevailing legal regulations.

BRP-Rotax GmbH & Co KG cannot accept any responsibility for the quality of work performed in accomplishing the requirements of this publication.

1.1) Applicability

For ROTAX® genuine spare parts for engine types:

Engine type	Serial number
916 iSc B	all
915 i A (Series)	all
915 i B (Series)	all
912 i (Series)	all
912 (Series)	all
914 (Series)	all
582	all

1.2) Concurrent ASB/SB/SI and SL

In addition to this Service Letter the following Service Bulletin/Service Instructions must be observed and complied with:

- SB-505-010/SB-535-009, title "New maintenance schedule, checklist and engine preservation", current issue.
- SI-11-1996, title "Engine preservation (2-stroke UL engines)", current issue.

1.3) Reason

- General information about storage and storage conditions of ROTAX® genuine spare parts
- Information on the shelf life for genuine spare parts used on ROTAX® aircraft engine types
- For storage of the engine control unit (ECU) of ROTAX® 912 iS / 912 iS Sport / 912 iSc Sport / 915 iS A / 915 iSc A / 915 iSc B / 916 iSc B special storage conditions are required by the manufacturer of the ECU.

NOTE: This Service Letter does not contain information on genuine spare parts made from silicone, Teflon, fluorocarbon or fluorosilicone which have a shelf life of more than 10 years.

1.4) Subject

Information about storage and storage conditions of ROTAX® genuine spare parts for ROTAX® aircraft engines.

SERVICE LETTER

1.5) Compliance

- None - for the implementation of this Service Letter.
- For storage conditions/engine preservation/storage and storage period of the engine observe the requirements of the latest valid "Instructions for Continued Airworthiness" for ROTAX® aircraft engines and the special requirements for ROTAX® spare parts indicated in this Service Letter.

1.6) Approval

The technical content of this document is approved under the authority of the DOA ref. EASA.21J.048.

1.7) Labor time and credit

None.

1.8) Mass data

Change of weight - - - none.

Moment of inertia - - - unaffected.

1.9) Electrical load data

No change.

1.10) Software modifications

No change.

1.11) References

In addition to this technical information refer to current issue of

- Operators Manual (OM)
- Illustrated Parts Catalog (IPC)
- Installation Manual (IM)
- Maintenance Manual Line (MML)
- Maintenance Manual Heavy (MMH)

NOTE: The status of the Manuals can be determined by checking the table of amendments. The 1st column of this table shows the revision status. Compare this number to the one listed on the ROTAX website:
www.flyrotax.com. Updates and current revisions can be downloaded for free.

1.12) Other Publications affected

None.

1.13) Interchangeability of parts

- Not affected

SERVICE LETTER

2) Material Information

2.1) Material

Price and availability will be provided on request by ROTAX® Authorized Distributors or their independent Service Centers.

2.2) Company support information

None.

2.3) Material requirement and credit per engine

None.

2.4) Material requirement and credit per spare part

None.

2.5) Rework of parts

None.

2.6) Special tooling/lubricants- /adhesives- /sealing compounds

Price and availability will be supplied on request by ROTAX® Authorized Distributors or their independent Service Centers:

Description	Qty/ engine	Part no.	Application
Service wiring harness	1	864280	Tool for reading the ECU.

SERVICE LETTER

3) Accomplishment/Instructions

- ROTAX® reserves the right to make any amendments to existing documents, which might become necessary due to this standardization, at the time of next revision or issue.

3.1) Instructions

3.1.1) General recommendation for the storage of spare parts

- Ideal storage temperature: 15 - 25 °C (59 °F - 77 °F)
- Dry storage (40 - 60% relative humidity)
- UV light-protected storage
- Ozone-free storage (no fluorescent light sources, mercury vapor lamps, printers etc.)
- Do not store directly on the floor, on walls, in front of radiators or other heat sources
- Dust-free storage (use the dust cover)
- Rust protection must not be removed (for example the oil film on untreated steel parts or the specially coated packaging materials)
- The storage facility will be protected from moisture, swelling agents such as fuels, oils or solvents and corrosive vapors.

NOTE: The conservation requirements and time limits of engines and components have to be observed and followed according to the specifications by the "Instructions for Continued Airworthiness".

3.1.2) Shelf life

The shelf life is the length of time a part can be stored before being installed on an aircraft (Not to be confused with SERVICE LIFE).

SERVICE LIFE: Service life is that period from installation to retirement.

3.1.3) ROTAX genuine spare parts

Shelf life on genuine spare parts does mainly affect parts made of rubber (like hoses, O-rings, oil seals,....). Rubber parts sent from BRP-Rotax GmbH & Co KG have a minimum of 48 months of shelf life beyond the invoice date.

For all other parts general storage requirements apply and also the relevant visual checks etc. apply when selling the parts and/or installing the part.

3.1.4) Special procedures with storage of the Engine Control Units (ECU)

Part	Period of storage time	Procedure
ECU	Up to 2 years	Visual inspection of the ECU before installation for traces of humidity and corrosion. If humidity is observed, the unit should be dried for 24 hours at low humidity and temperatures from 70-80 °C (158 - 176 °F). At signs of corrosion (at the connector pins) these ECUs should be scrapped.
	At longer than 2 years	The ECU should be activated for approx. 30 min and be operated. At signs of corrosion (at the connector pins) these ECUs should be scrapped. Reference Maintenance Manual Heavy Chapter 72-10-00.

d06771.fm

SERVICE LETTER

There are three different scenarios where the ECU is stored or installed:

See [Fig. 1](#).

- a) as a spare part ECU stored in a shelf (to be activated on engine test bench or via service harness)
- b) as an ECU provided together with an engine in the crate (to be activated on engine test bench or via service harness separate from the engine)
- c) as an ECU not operated installed in an aircraft for a certain time period.

3.1) Test run

Not applicable.

3.2) Summary

These instructions (section 3) have to be followed in accordance with the deadlines specified in section 1.5.

The execution of the Service Letter must be confirmed in the logbook.

! A revision bar outside of the page margin indicates a change to text or graphic.

Translation into other languages might be performed in the course of language localization but does not lie within ROTAX® scope of responsibility.

In any case the original text in English language and the metric units are authoritative.

3.3) Inquiries

Inquiries regarding this Service Letter should be sent to the ROTAX® Authorized Distributor of your area.

A list of all ROTAX® Authorized Distributors or their independent Service Centers is provided on www.flyrotax.com.

4) Appendix

The following drawings should convey additional information:

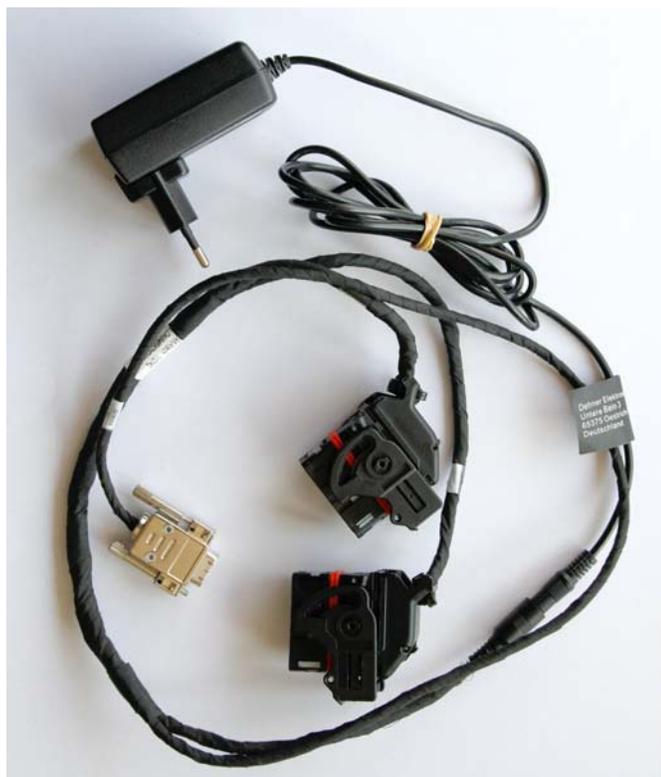


Fig. 1
Service wiring harness

10124

NOTE: The illustrations in this document show the typical construction. They may not represent full detail or the exact shape of the parts which have the same or similar function.
Exploded views are **not technical drawings** and are for reference only. For specific detail, refer to the current documents of the respective engine type.